

# 一种基于 DSMT 和 HMM 的序列飞机目标识别算法

李新德<sup>1</sup> 潘锦东<sup>1</sup> DEZERT Jean<sup>2</sup>

**摘要** 提出一种基于 DSMT (Dezert-Smarandache theory) 和 HMM (Hidden Markov model, HMM) 的序列飞机目标识别算法 (Multiple features and sequential information fusion, MFSIF). 该算法首先对序列图像进行二值化处理, 然后提取 Hu 矩和轮廓的局部奇异值. 其次, 使用 PNN (Probabilistic neural networks, PNN) 构建 BBA (Basic belief assignment, BBA); 最后, 利用 DSMT 融合不同特征, 得到 HMM 的观察序列. 通过计算观察序列与每个 HMM 的相似度, 完成飞机目标的识别. 实验结果表明, 该算法具有较高的识别率, 且当连续遮挡帧数  $\zeta \leq 6$  时, 仍能保持较高的识别率.

**关键词** 序列飞机, 目标识别, DSMT, HMM, PNN, BBA, MFSIF.  
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## A Target Recognition Algorithm for Sequential Aircraft Based on DSMT and HMM

LI Xin-De<sup>1</sup> PAN Jin-Dong<sup>1</sup> DEZERT Jean<sup>2</sup>

**Abstract** For the low rate problem of recognition of automatic target recognition of aircraft caused by the great change of posture, a target recognition algorithm based on DSMT (Dezert-Smarandache theory) and HMM (Hidden Markov model) is proposed by utilizing the fusion of multiple features and sequential information (i.e. MFSIF). The novelty of the algorithm is integrating the multiple feature fusion recognition with the sequential images fusion recognition. Firstly, the sequential images are preprocessed with the binarization method, then the features of Hu moments and partial singular values of outline are picked up. Secondly, the PNN (Probabilistic neural network) is used to construct the BBA (Basic belief assignment). After that, these different features from the same image are fused by DSMT in order to gain the observation sequence of HMM. Then, the sequential information is fused by utilizing HMM to finish the automatic recognition of aircraft with varied multiple postures by calculating the similarities between the observation sequence and each HMM. Finally, according to the simulation experiment results, this algorithm has a high exact recognition rate even for aircraft with greatly varied postures. Simultaneously, this algorithm can also satisfy the requirement of aircraft target recognition in real-time. In addition, this algorithm can also guarantee a high recognition rate of aircraft target when the number  $\zeta$  of consecutive occulted frames is less than or equal to six.

**Key words** Sequential aircrafts, target recognition, multiple features fusion, Dezert-Samarandache theory (DSMT), probabilistic neural network (PNN), sequential information fusion, hidden Markov model (HMM)

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1. Key Laboratory of Measurement and Control of CSE of  
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(Automatic target recognition, ATR)  
(Automatic aircraft recognition, AAR) ATR

Ministry of Education, School of Automation, Southeast University, Nanjing 210096, China 2. The French Aerospace Lab, Palaiseau F-91761, France

[1, 3] ATR  
 [4, 7] [8]  
 BP  
 (Multiple features automatic target recognition, MF-ATR)<sup>[9, 13]</sup>  
 [12] BP DST (Dempster-Shafer theory)  
 [13] Hu

(Probabilistic neural networks, PNN)  
 DSmT (Dezert-Smarandache theory)

[14, 15] [14]  
 Hu BP  
 DST Hu  
 [15] DSmT  
 BP [14, 15]

DSmT  
 (Hu  
 DSmT PNN  
 (Hidden Markov model, HMM)  
 HMM HMM

### 1 序列飞机目标识别框架

DSmT HMM  
 1  
 T  
 :  
 1) Hu  
 ;  
 2) DSmT  
 PNN  
 (Basic belief assignment, BBA).  
 DSmT  
 3) :  
 HMM

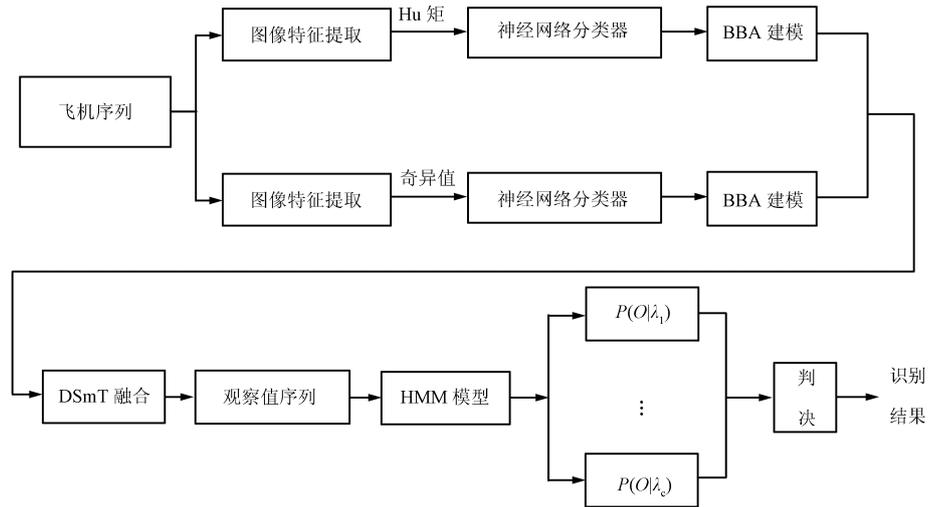
#### 1.1 图像特征提取

##### 1.1.1 Hu 矩

$f(x, y), (p + q)$   
 $m_{pq} (p + q) \mu_{pq}$  :

$$m_{pq} = \sum_{m=1}^M \sum_{n=1}^N m^p n^q f(m, n) \quad (1)$$

[7, 16]  
 [17]



1 DSmT HMM

Fig.1 Target recognition algorithm of sequential aircrafts based on DSMT and HMM

$$\mu_{pq} = \sum_{m=1}^M \sum_{n=1}^N (m \text{ i } \bar{x})^p (n \text{ i } \bar{y})^q f(m, n) \quad (2)$$

$$\eta_{pq} = \frac{\mu_{pq}}{\mu_{0,0}^r}, r = \frac{p+q}{2}, p+q = 2, 3, \dots \quad (3)$$

(4)  $\eta_{pq}$

$$\begin{aligned} \textcircled{1} &= \eta_{20} + \eta_{02} \\ \textcircled{2} &= (\eta_{20} \text{ i } \eta_{02})^2 + 4\eta_{11}^2 \\ \textcircled{3} &= (\eta_{30} \text{ i } 3\eta_{12})^2 + (3\eta_{21} \text{ i } \eta_{03})^2 \\ \textcircled{4} &= (\eta_{30} + \eta_{12})^2 + (\eta_{21} + \eta_{03})^2 \\ \textcircled{5} &= (\eta_{30} \text{ i } 3\eta_{21})(\eta_{30} \text{ i } \eta_{21})[(\eta_{30} + \eta_{12})^2 \text{ i } \\ &\quad 3(\eta_{21} + \eta_{03})^2] + (3\eta_{21} \text{ i } \eta_{03})(\eta_{21} + \eta_{03}) \text{E} \\ &\quad [3(\eta_{30} + \eta_{12})^2 \text{ i } (\eta_{21} + \eta_{03})^2] \\ \textcircled{6} &= (\eta_{20} \text{ i } \eta_{02})[(\eta_{30} + \eta_{12})^2 \text{ i } (\eta_{21} + \eta_{03})^2] + \\ &\quad \eta_{11}(\eta_{30} + \eta_{12})(\eta_{03} + \eta_{21}) \\ \textcircled{7} &= (3\eta_{21} \text{ i } \eta_{03})(\eta_{30} + \eta_{12})[(\eta_{30} + \eta_{12})^2 \text{ i } \\ &\quad 3(\eta_{03} + \eta_{21})^2] + (3\eta_{12} \text{ i } \eta_{30})(\eta_{21} + \eta_{03}) \text{E} \\ &\quad [3(\eta_{30} + \eta_{12})^2 \text{ i } (\eta_{03} + \eta_{21})^2] \end{aligned} \quad (4)$$

$$\textcircled{1}, \textcircled{2}, \textcircled{3}, \textcircled{4}$$

1.1.2 轮廓局部奇异值

定义 1.  $A_{m \in n}$

$$U^H A V = \begin{pmatrix} S & 0 \\ 0 & 0 \end{pmatrix} \quad (5)$$

$S = \text{diag}\{\sigma_1, \sigma_2, \dots, \sigma_r\}$ ,  $\sigma_1, \sigma_2, \dots, \sigma_r > 0$ ,  $\sigma_i (i = 1, 2, \dots, r)$

$$\sigma_i = \sqrt{\frac{\lambda_i}{\lambda_i}}$$

[19]

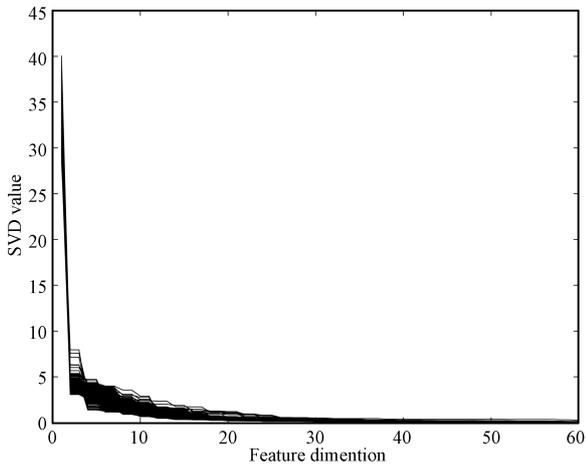


Fig. 2 Distribution diagram of global singular values of outline

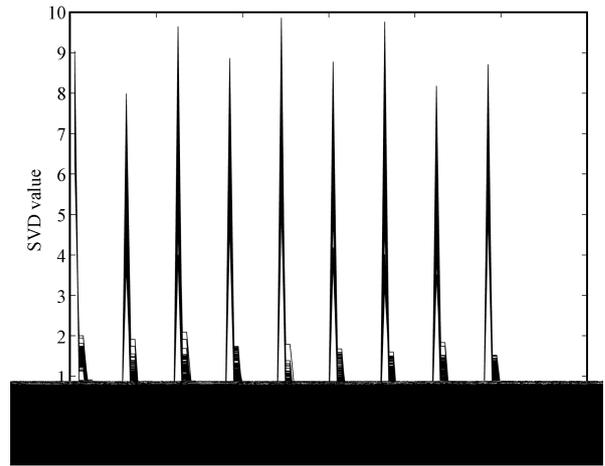


Fig. 3 Distribution diagram of partial singular values of outline

$$[d_1, d_2, \dots, d_n].$$

$$[d_1, d_2, \dots, d_n]$$

$$l = \frac{n}{w} + 1 \quad (6)$$

$l \in w$

200

3

3

3

1.2.1 DSmT 建模

DSmT Dezert Smarandache [20] [23]

DST

$$D^E$$

$$\theta_1, \theta_2, \dots, \theta_n$$

$$m_i$$

(BBA),

$$m_i(\theta_j)$$

DSmT

$$S_1 \quad S_2$$

(7) [24]:

$$m_{\mu^f(E)}(X) \cdot m(X) = \sum_{\substack{\theta_i, \theta_j \in D^E \\ \theta_i \setminus \theta_j = X}} m_1(\theta_i) m_2(\theta_j) \quad (7)$$

1.2 多特征融合

Hu

DSmT

signment, BBA),

DSmT

DSmT

BBA

PNN

(Basic belief as-

PCR5

PNN

PCR (Proportional conflict redistribution)<sup>[24]</sup>

PCR

PCR6

PCR6

PCR5,

1 DSMT  
Table 1 The modeling of DSMT

	$S_1$	$S_2$	...	$S_k$
$\theta_1$	$m_1(\theta_1)$	$m_2(\theta_1)$	...	$m_k(\theta_1)$
$\theta_2$	$m_1(\theta_2)$	$m_2(\theta_2)$	...	$m_k(\theta_2)$
$\vdots$	$\vdots$	$\vdots$	$\ddots$	$\vdots$
$\theta_n$	$m_1(\theta_n)$	$m_2(\theta_n)$	...	$m_k(\theta_n)$

$$\begin{cases} X = ; , m_{PCR5}(X) = 0 \\ 8X \ 2 \ D^{\text{inf}}; g, m_{PCR5}(X) = m_{12}(X) + \\ \sum_{\substack{Y \ 2 \ D^{\text{inf}} \\ X \setminus Y = ;}} \left[ \frac{m_1(X)^2 m_2(Y)}{m_1(X) + m_2(Y)} + \frac{m_2(X)^2 m_1(Y)}{m_2(X) + m_1(Y)} \right] \end{cases}$$

(8)

$$m_{12}(X) \cdot m_{\setminus}(X) \quad (7)$$

(8)

3)  $c \in k$   $y$   $Hu$   $PNN$   $(Learning \ vector \ quantization, \ LVO)$   $DSMT$

1.2.2 BBA 的 PNN 赋值 DSMT

Bayesian BBA

$$m_{ij}^0 = \frac{m_{ij}}{\sum_{j=1}^y m_{ij}}, \quad i = 1, 2 \quad (9)$$

Specht

[25i 26]

PNN

(9)

DSMT

PNN

PCR5

1.3 序列信息融合

PNN

[13] PNN

DSMT

BBA.

PNN

BBA

$c$

$k$

$y$  ( DSMT

$N$  ,  $S = \{s_1, s_2, \dots, s_N\}$

1)

$Hu$

$Q = (q_1, q_2, \dots, q_T)$

$S$

$t$

$q_t$

$t$

$s_j$

$1, 2, \dots, t_j - 1$

2)

$t$

$q_t$

$$P(q_t = s_j | q_{t-1} = s_i, q_{t-2} = s_k, \dots) = \frac{P(q_t = s_j | q_{t-1} = s_i)}{P(q_t = s_j | q_{t-1} = s_i)} \quad (10)$$



4

Fig. 4 The state of aircraft

(Markov chain).

1)  $N$ :

$$S = \{s_1, s_2, \dots, s_N\}, \quad t = 1 \dots T$$

2)  $\pi$ :

$$\pi_i = P(q_1 = s_i), \quad 1 \leq i \leq N \quad (11)$$

3)  $M$ :

$$V = \{v_1, v_2, \dots, v_M\}, \quad t = 1 \dots T$$

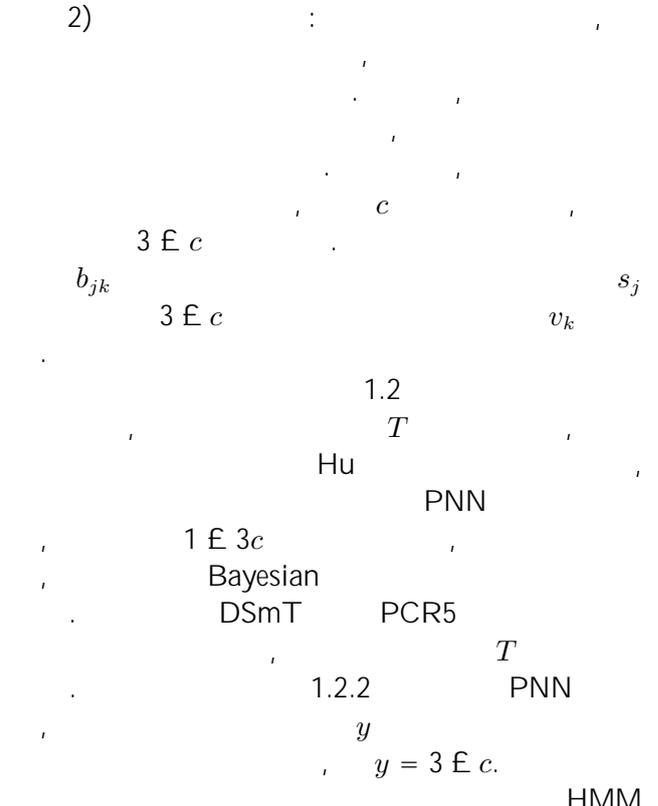
4)  $A$ :

$$A = \{a_{ij} | i, j \in N\}, \quad a_{ij} = P(q_{t+1} = s_j | q_t = s_i) \quad (12)$$

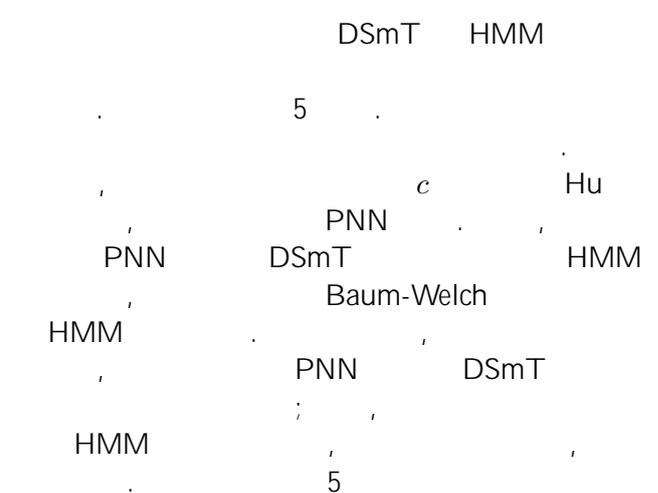
5)  $B$ :

$$B = \{b_{jk} | j \in N, k \in M\}, \quad b_{jk} = P(o_t = v_k | q_t = s_j) \quad (13)$$

$$\lambda = (A, B, \pi) \quad (14)$$

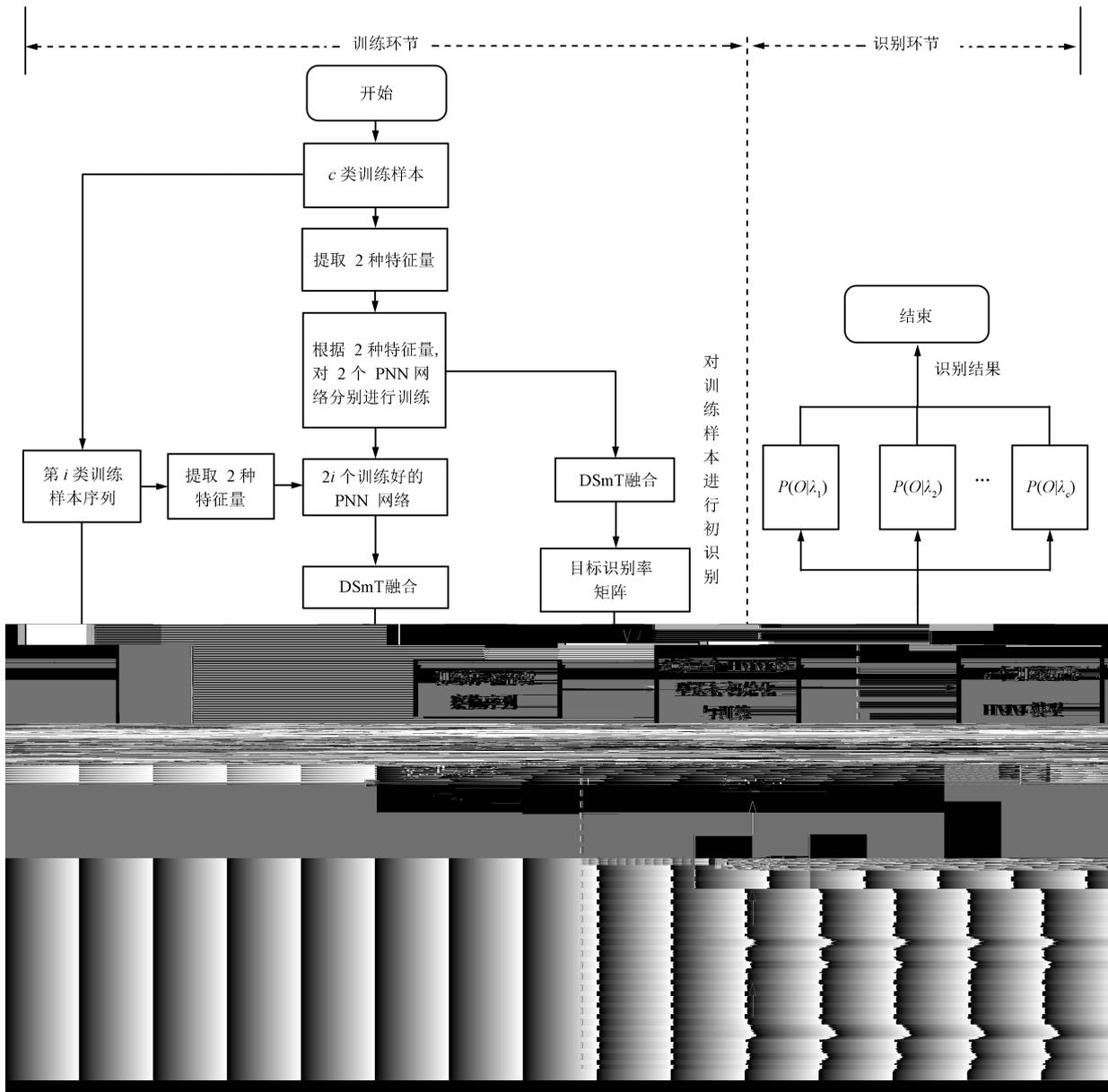


2 算法流程



2.1 HMM 模型训练

HMM  $\lambda = (A, B, \uparrow)$   
 HMM Baum-Welch  
 $P(O|\lambda)$   
 (Maximization)<sup>[34]</sup>  
 Baum-Welch  
 A B



5 DSMT HMM

Fig. 5 Flowchart of the target recognition of sequential aircrafts based on DSMT and HMM algorithm

- 
- 1)  $A$  .  
' ; , ' ; ;  $\frac{1}{2}$

$P(O_T | \lambda_i), i = 1, \dots, c.$  HMM  $O(c^2),$   
 (Bayesian HMM  $O(c^3).$  , MFSIF  $O(c^3) + O(c^2),$  , MFSIF  $O(c^3).$   
 $class = \arg \max_{1 \leq i \leq c} P(O | \lambda_i)g$  (20)  $O(c^2).$

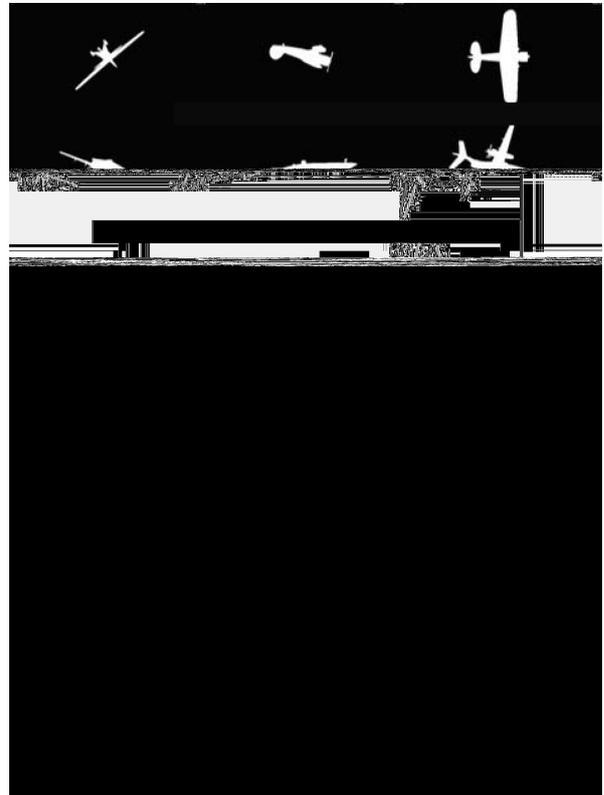
3 实验结果与分析

DELL OPTIPLEX380 PC  
 Visual Studio 2008  
 CPU: Pentium (R) Dual-Core E5500  
 @2.80 GHz 2.80 GHz, : 2 GB, : 32  
 Windows 7 7

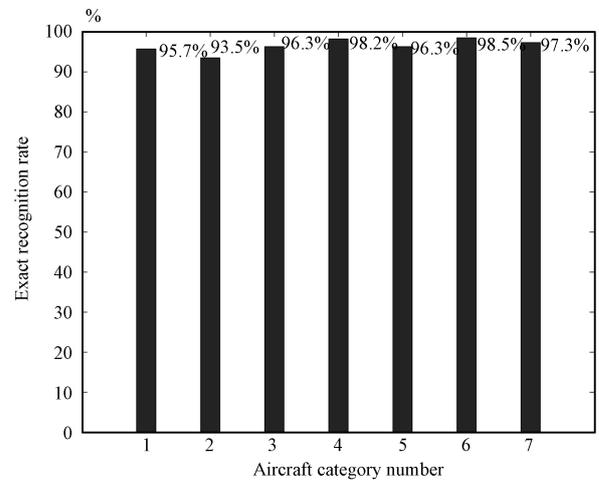
150 (7  
 1 200 £ 702).  
 6  
 7 7  
 400 » 500 PNN 1.2.2  
 y  
 PNN  
 2 PNN  
 ( ) 0.1.

3.1 实验 1: 正确识别率与实时性

DSMT HMM  
 (Multiple features and se-  
 quential information fusion, MFSIF),  
 7  
 93.5%  
 MFSIF  
 c ,



6 7  
 Fig. 6 Observations of 7 classes of aircraft



7  
 Fig. 7 Exact recognition rate for every kind of aircraft

8 î •, ' ÓØE MFSIF ¥9 Ø<sup>-</sup> ' L 1 ðØE (MFF1 aMFF2 aSF1 „ SF2) ÚB Ä, MYHW r q µ t †®. ØEY V v L V ü, MFSIF ' L v v 4 Ú ØE¥ž' MYq „ Zÿ Ÿ, Oáì YVL ©k, ÿi ž Ä-m^¥MYH W 5»7ms - W, › †j @f " SMYØE¥ LHÿ1 p.

3.2 L 2: • " 3 g MÄL

' L Ì, p " ' o „ ©k " ' o ÐL 1 M]. ö 1 l ³ • " 3 g É MÄH MFSIF ØE ¥ž' MYq. 1 £ • " a x ¥ + ~ ' ? ÷ %Á¥í • } •, f Úxβ3 g É m ¥ | ' < ¾ { l ¿ • " 3 g É w. 6 B Z ë, • " 3 g É < ù Bç S ¶ =, y 1 ÂT • " 3 g É V v, î 4 | ¥ s ' \_ 5 ö C " ¥ , ' , • Á + ~ µ r ' , , í EW %' í • } • ' É. 7 • " 3 g V l , 5 ° ? í • % « + ~ , 7 } • ¥ œ8 + ~ í • , . » y 1 , } • † s ' ¥ + ~ ' B î 6 » [ a ö á í š h. î [ , ' L Ì, • " 3 g É K I É 4 1 6. V 2 ó , ] • " 3 g É Ð x β 3 g É 1 ' / MFSIF ¥ž' MY q.

L ² Ts : Y V V 2 V [ ? C, ' • " 3 g v ¿ 14 H, ž' MYq Û " • " 3 g ¥ 9 F ) ¿ / Á ¥ t ] . ' • " 3 g 9 F ž 18 H ÷ F ü A. î [ , ' • " 3 g ! ç S ¶ 6 » 14 = H, ' x β 3 g @ v ¥ f f / , MFSIF ØE ¥ž' MY q, ù È ¿ • " 3 g ¥ MÄ.

3.3 L 3: j ê b L

' L Ì, l ³ Û " " S ê l H, MFSIF ØE ^ Ž ? ž' µ r ¥ " S É › MY, ' l ³ MFSIF ØE " S ê l H ¥ Z ÿ Ÿ. p " ' o ÐL 1 M], ©k o ^ ð Ÿ ©k o

¥ \$ , S B ç 1 è ê b, s Y ê l À ð Ÿ 1/2 a1/4 a1/8 a1/16 a1/32 a1/64. l ³ NH " S ¥ ž' MY q. MY² T Â m 8 î U.

L ² Ts : ©m 8 V [ A , ' f " S ? 3 ê b H, p È Ì Ä B È ¥ MY ž' q ù B ç ¥ - = ? 3 l S ¶ œ î , ' P " S ê l ž ð Ÿ ¥ 1/8, 9 V [ £ 90 % P . ¥ ž' MY q, V n " S B ç ñ ê l H, ' Z E 9 V [ £ " z ¥ Z ÿ Ÿ. ØE " S ê l ž ð S j ¥ 1/16 H, ØE ¥ž' MY q ? 3 1 ü A ¥ / Á. 7 ' " S É B „ ê l ž 1/32 a1/64 H, K ® MY q s Y / † À 36.70 % „ 19.20 %, NH ØE ' > r .

3.4 L 4: " S ¿ ML

É › f MYH, î ³ 1 MY ¥ " S V ? ö y 1 H W ¥ w M a ð " S \$ ` î © " , y í 7 ? 3 ¿ M, ' Ó 4 MFSIF ØE " ¿ + ~ • ¥ µ † - " , î ¿ ½ • ¥ µ † É › " S M Y , î [ 1 £ ' î M Y " S ? 3 ¿ M H , ^ Ž y 1 V ¿ G ¿ ½ • 7 • Á v S ¶ ¥ M Y p . ' L ¥ p " ' o ÐL 1 M], ©k " ' o • " ñ ©k " ' ½ . ©k " ' ½ A ^ ® 2 ñ È Y f F î ¥ † È Y ¥ ©k " ' ½ . ©k " ' ½ A ® 857 f m È F î . Ì È Y 1 ¥ f m È F î - 475 m, È Y 2 f m È F î a 382 . ©k " ' ½ B ^ ® 3 È f F î ¥ È È Y ©k " ' ½ . ©k " ' ½ B ©k " ' ½ A ¥ \$ @ F B È f F î ¥ • ¥ ©k " ' ½ , ' ® 1340 m È F î , - 475 ® È Y 1 ¥ f m È F î , Ì W ¥ 382 m È ® È Y 2 ¥ f m È F î , a 487 m È ® È Y 3 ¥ f m È F î . d 9 f ñ ©k " ' " † N a N a È È 1 5 7 5 > P J F 1 10.5

3.5 实验 5: 与基于多特征融合算法对比试验

MFSIF  
 MFSIF (Normalized  
 Hu  
 moment of inertia, NMI)  
 DSMT  
 1-MFF1) Hu NMI [13] (  
 DST  
 2-MFF2) [12] (  
 MFF1 PNN  
 BP  
 MFF2  
 BP  
 S  
 Levenberg-Marquardt  
 $nodes = \frac{p}{ts} + d, 1 \cdot d \cdot 10$   
 0.05,  
 t, s

8

Fig. 8 Exact recognition rate when images zoom out

Table 3 The exact recognition rate of test sample sequence A and the composite type (%)

A (%)	
1	2
98.5	96.3
97.3	

Table 4 The exact recognition rate of test sample sequence B and the composite type (%)

B (%)			
1	2	3	4
98.5	96.3	96.3	96.8

MFSIF  
 97.3%.  
 98.5% 96.3%.  
 A  
 97.5%.  
 B, MFSIF 96.8%.  
 98.5% 96.3% 96.3%.  
 B  
 A 97.1%.  
 MFSIF B

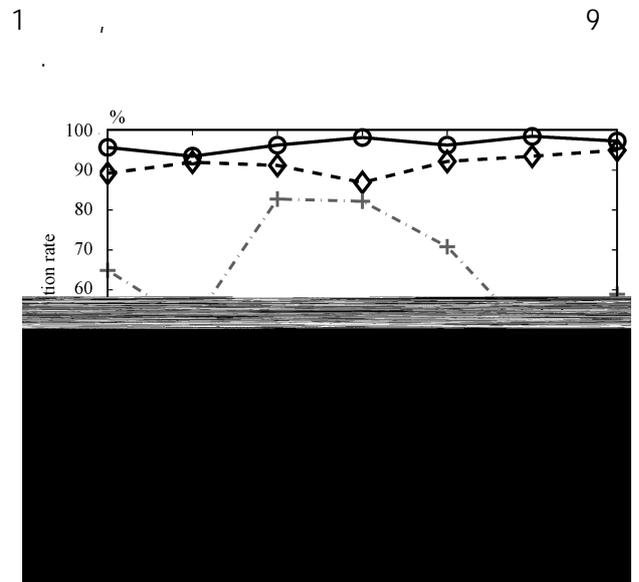


Fig. 9 Comparison of exact recognition rate with multiple features fusion algorithms

MFSIF  
 93.5%,  
 MFF1  
 MFF2

MFSIF, 7, BP, BP, MFF2, 48.3%, 82.75%, 60%

3.6 实验 6: 与基于单一特征的序列融合算法对比试验

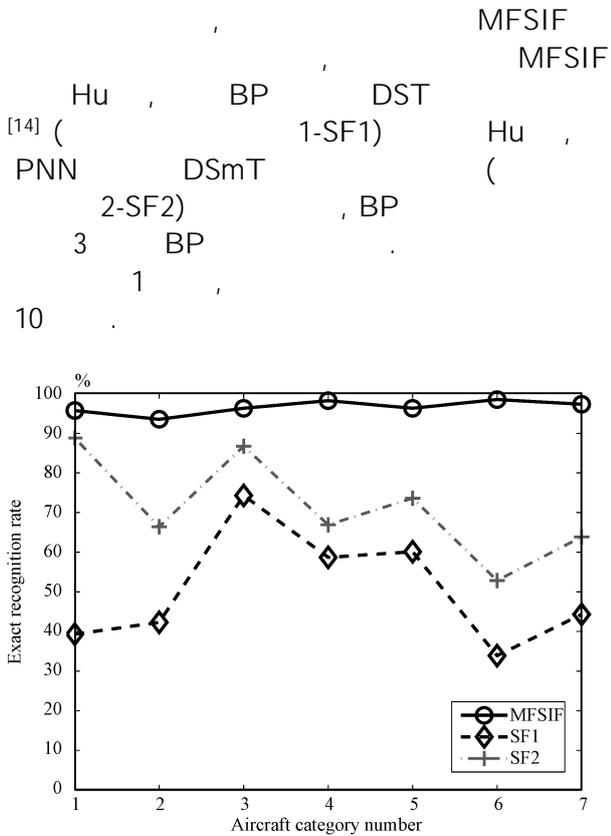


Fig. 10 Comparison of exact recognition rate with sequential fusion algorithms

3.7 实验 7: 遮挡实验

$$L_s = 475$$

$L_w = 7$   
(Monte-Carlo simulation)  
 $\tau \cdot 6$   
 $N = 500$

$$T_w = T_o + T_n$$

$$T_a = \frac{T_w}{N}$$

$$AR_o = 1 - \frac{L_w + T_a}{L_s}$$

$$AR_n = 1 - \frac{T_a}{(L_s - L_w)}$$

$$T_w = T_o + T_n \tag{21}$$

$$T_a = \frac{T_w}{N} \tag{22}$$

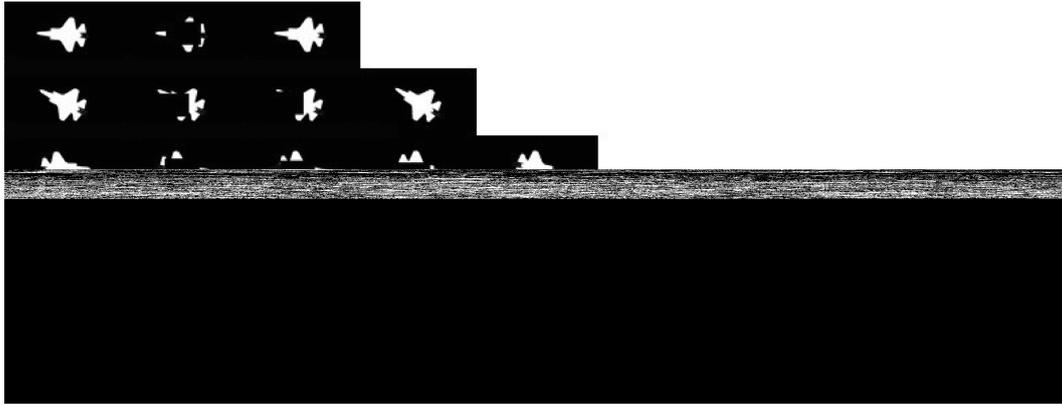
$$AR_o = 1 - \frac{L_w + T_a}{L_s} \tag{23}$$

$$AR_n = 1 - \frac{T_a}{(L_s - L_w)} \tag{24}$$

5, 1 > 6, MF-  
SIF, AR\_o = 98.13%  
AR\_n = 99.60%

4 结束语

MFSIF, DSMT, HMM, SF1, SF2, MFSIF, MFSIF



11

Fig. 11 Partial occlusion of aircraft

5

Table 5 Test results under different occulted conditions

	$T_o$	$T_n$	$T_w$	$T_a$	$AR_o$	$AR_n$
1	0	0	0	0	98.53 %	100 %
2	84	9	93	0.186	98.49 %	99.96 %
3	248	24	272	0.544	98.41 %	99.88 %
4	383	48	431	0.862	98.34 %	99.82 %
5	593	74	667	1.334	98.25 %	99.71 %
6	819	127	946	1.892	98.13 %	99.60 %

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李新德

E-mail: xindeli@seu.edu.cn

(**LI Xin-De** Associate professor at the School of Automation, Southeast University. His research interest covers intelligent robot, human-robot interaction, robot perception, information fusion, reasoning under uncertainty, and machine vision. Corresponding author of this paper.)

潘锦东  
2014

E-mail: panjindong1989@163.com

(**PAN Jin-Dong** Master student at the School of Automation, Southeast University. He received his master degree from Southeast University in 2014.

His research interest covers information fusion and image processing.)



DEZERT Jean

E-mail: jdezert@gmail.com

(**DEZERT Jean** Senior scientist in ONERA (The French Aerospace Lab). His research interest covers information fusion, reasoning under uncertainty, auto navigation, multi-sensor and multi-object tracking.)